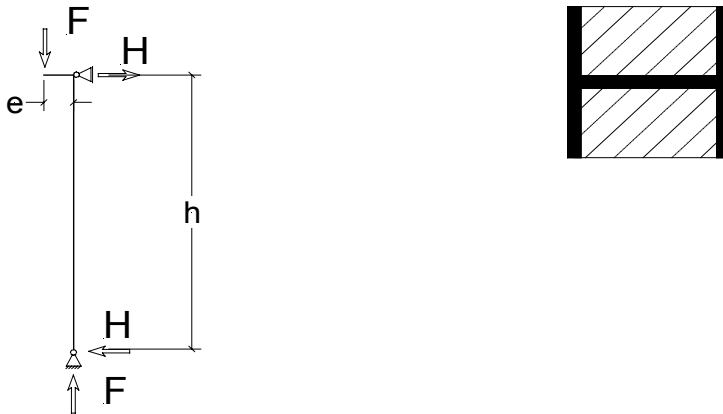


Euro-Code 4

Columns

Concrete encasement composite column with eccentric force application:



System:

Height h =	8,00 m	
Eccentricity e =	0,15 m	
Beam type =	SEL("steel/profils"; Name;)	= HEB
Nominal height NH =	SEL("steel/type; NH;)	= 340
A_a =	TAB("steel/type; A; NH=NH)	= 171,00 cm ²
h_p =	TAB("steel/type; h; NH=NH)	= 340,00 mm
s =	TAB("steel/type; s; NH=NH)	= 12,00 mm
t =	TAB("steel/type; t; NH=NH)	= 21,50 mm
b =	TAB("steel/type; b; NH=NH)	= 300,00 mm
I_{ya} =	TAB("steel/type; I_y ; NH=NH)	= 36660,00 cm ⁴
W_{pa} =	1.14 * TAB("steel/type; W_y ; NH=NH)	= 2462,40 cm ³
W_{pa} =		2408,00 cm ³

Materials:

Concrete =	SEL("concrete/EC"; Name;)	= C30/37
Steel =	SEL("steel/EC"; NameEN;)	= S235
E_{cm} =	TAB("concrete/EC"; E_{cm} ; Name=Concrete)	= 32000,00 N/mm ²
f_{ck} =	TAB("concrete/EC"; f_{ck} ; Name=Concrete)	= 30,00 N/mm ²
α =	0,85	
E_a =	TAB("steel/EC"; E; NameEN=Steel)	= 210000,00 N/mm ²
f_{yk} =	TAB("steel/EC"; f_y ; NameEN=Steel)	= 235,00 N/mm ²

Partial safety factors:

Dead load γ_G =	1,35
Imposed Load γ_Q =	1,50
Concrete γ_c =	1,50
Construction steel γ_a =	1,10
Profile steel sheeting γ_{ap} =	1,10
Longitudinal shear γ_{vs} =	1,25
Elastic modulus γ =	1,35

Load:

$$g_k = 450,00 \text{ kN}$$

$$q_k = 1000,00 \text{ kN}$$

Calculation:

$$N_{Sd} = g_k * \gamma_G + q_k * \gamma_Q = 2107,50 \text{ kN}$$

$$M_{Sd} = N_{Sd} * e = 316,13 \text{ kNm}$$

$$V_{Sd} = M_{Sd} / h = 39,52 \text{ kN}$$

$$f_{cd} = \alpha * f_{ck} / \gamma_c = 17,00 \text{ N/mm}^2$$

$$f_{yd} = f_{yk} / \gamma_a = 213,64 \text{ N/mm}^2$$

$$E_{cd} = E_{cm} / \gamma = 23703,70 \text{ N/mm}^2$$

$$A_c = (h_p * b) / 100 - A_a = 849,00 \text{ cm}^2$$

$$I_{yc} = (h_p^3 * b / 120000) - I_{ya} = 61600,00 \text{ cm}^4$$

Local buckling of parts of the steel section:

$$\epsilon = \sqrt{(235 / f_{yk})} = 1,00$$

$$(b / t) / (\epsilon * 44) = \underline{\underline{0,32 < 1}}$$

Load-bearing capacity of the column under centric pressure with buckling risk:

$$N_{pl,Rd} = (A_a * f_{yd} + A_c * f_{cd}) / 10 = 5096,54 \text{ kN}$$

Structural verification procedure applicable?

$$\delta = A_a * f_{yd} / (10 * N_{pl,Rd}) = 0,72$$

$$0,2 / \delta = \underline{\underline{0,28 < 1}}$$

$$\delta / 0,9 = \underline{\underline{0,80 < 1}}$$

⇒ Structural verification procedure applicable.

On the effect of lateral shear:

$$A_v = (1,04 * h_p * s) / 100 = 42,43 \text{ cm}^2$$

$$V_{pl,Rd} = A_v * f_{yd} / (\sqrt{3} * 10) = 523,35 \text{ kN}$$

$$V_{Sd} / V_{pl,Rd} = \underline{\underline{0,08 < 0,5}}$$

⇒ The lateral shear force has no effect on the load-bearing capacity.

Cross section bearing capacity under pressure and single-axis bending:

Plastic limit moment at point D:

$$W_{pc} = b * h_p^2 / 4000 - W_{pa} = 6262,00 \text{ cm}^3$$

$$M_{max,Rd} = (W_{pa} * f_{yd} + 1/2 * W_{pc} * f_{cd}) / 10^3 = 567,67 \text{ kNm}$$

$$N_D = 1/20 * A_c * f_{cd} = 721,65 \text{ kN}$$

Plastic limit moment in points B and C:

$$N_{pm,Rd} = (A_c * f_{cd}) / 10 = 1443,30 \text{ kN}$$

$$h_N = N_{pm,Rd} / (0,2 * b/10 * f_{cd} + 2 * s/100 * (2 * f_{yd} - f_{cd})) = 7,20 \text{ cm}$$

$$h_N / ((b / 2 - t) / 10) = \underline{\underline{0,56 < 1}}$$

⇒ Plastic neutral axis is in the rib!

$$W_{pan} = s/10 * h_N^2 = 62,21 \text{ cm}^3$$

$$W_{pcn} = (b * h_N^2) / 10 - W_{pan} = 1492,99 \text{ cm}^3$$

$$M_{N,Rd} = (W_{pan} * f_{yd} + 1/2 * W_{pcn} * f_{cd}) / 10^3 = 25,98 \text{ kNm}$$

The plastic limit moment in points B and C is:

$$M_{pl,Rd} = M_{max,Rd} - M_{N,Rd} = 541,69 \text{ kNm}$$

$$N_B = 0,00 \text{ kN}$$

$$N_C = 2 * N_D = 1443,30 \text{ kN}$$

Load-bearing capacity of a composite column under pressure and single-axis bending:

$$N_{pl,Rk} = (A_a * f_{yk} + A_c * \alpha * f_{ck}) / 10 = 6183,45 \text{ kN}$$

Effective elastic bending rigidity:

$$EI_e = (E_a * I_{ya} + 0.8 * E_{cd} * I_{yc}) / 10^5 = 88667,18 \text{ kN/m}^2$$

Reference degree of slenderness:

$$N_{cr} = \pi^2 * EI_e / h^2 = 13673,57 \text{ kN}$$

$$\lambda' = \sqrt{(N_{pl,Rk} / N_{cr})} = 0,67$$

⇒ Long-term behaviour (creep and shrinkage) may be disregarded.

Strut curve b:

$$\text{Type} = \text{SEL}(\text{"comp/buck"; Desc; }) = \text{concrete-encased profile strong axis}$$

$$\text{line} = \text{TAB}(\text{"comp/buck"; line; Desc=Type}) = \text{b}$$

$$\kappa = \text{TAB}(\text{"comp/buck"; } \kappa; \text{line=line; } \lambda_{lat}=\lambda') = 0,800$$

$$\kappa_d = N_{Sd} / N_{pl,Rd} = 0,414$$

$$\text{Proportion of edge moments } r = 0,00$$

$$\kappa_N = \kappa * ((1 - r) / 4) = 0,200$$

The resulting values for point C are:

$$\kappa_c = N_c / N_{pl,Rd} = 0,283$$

$$\mu_c = 1,00$$

$$\mu_k = (1 - \kappa) / (1 - \kappa_c) * \mu_c = 0,279$$

$$\mu_d = (1 - \kappa_d) / (1 - \kappa) * \mu_k = 0,817$$

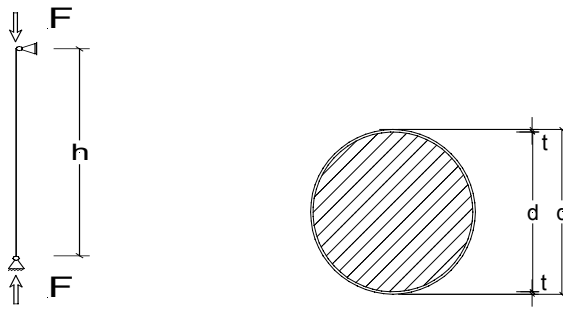
The length μ is:

$$\mu = \mu_d - \mu_k * (\kappa_d - \kappa_N) / (\kappa - \kappa_N) = 0,717$$

Structural verification of load-bearing capacity:

$$N_{Sd} / (\kappa * N_{pl,Rd}) = \underline{\underline{0,52 < 1}}$$

$$M_{Sd} / (0.9 * \mu * M_{pl,Rd}) = \underline{\underline{0,90 < 1}}$$

Concrete-filled hollow section:**System:**

Column height $h = 7,00 \text{ m}$
 Outside pipe diameter $d = 273,00 \text{ mm}$
 Pipe thickness $t = 6,30 \text{ mm}$

Load:

$g_k = 500,00 \text{ kN}$
 $q_k = 600,00 \text{ kN}$

Materials:

Concrete = SEL("concrete/EC"; Name;) = C30/37
 Steel = SEL("steel/EC"; NameEN;) = S355
 $E_{cm} = \text{TAB}(\text{"concrete/EC"; } E_{cm}; \text{ Name=Concrete}) = 32000,00 \text{ N/mm}^2$
 $f_{ck} = \text{TAB}(\text{"concrete/EC"; } f_{ck}; \text{ Name=Concrete}) = 30,00 \text{ N/mm}^2$
 $E_a = \text{TAB}(\text{"steel/EC"; } E; \text{ NameEN=Steel}) = 210000,00 \text{ N/mm}^2$
 $f_{yk} = \text{TAB}(\text{"steel/EC"; } f_y; \text{ NameEN=Steel}) = 355,00 \text{ N/mm}^2$
 $\alpha = 1,00$

Partial safety factors:

Dead load $\gamma_G = 1,35$
 Imposed Load $\gamma_Q = 1,50$
 Concrete $\gamma_C = 1,50$
 Construction steel $\gamma_a = 1,10$
 Profile steel sheeting $\gamma_{ap} = 1,10$
 Longitudinal shear $\gamma_{vs} = 1,25$
 Concrete elastic modulus $\gamma = 1,35$

Calculation:

$f_{yd} = f_{yk} / \gamma_a = 322,73 \text{ N/mm}^2$
 $f_{cd} = \alpha * f_{ck} / \gamma_C = 20,00 \text{ N/mm}^2$
 $N_{sd} = \gamma_G * g_k + \gamma_Q * q_k = 1575,00 \text{ kN}$
 $f_{yd} = f_{yk} / \gamma_a = 322,73 \text{ N/mm}^2$
 $A_a = \pi / 400 * (d^2 - (d - 2 * t)^2) = 52,79 \text{ cm}^2$
 $I_a = \pi / 640000 * (d^4 - (d - 2 * t)^4) = 4695,82 \text{ cm}^4$
 $E_{cd} = E_{cm} / \gamma = 23703,70 \text{ N/mm}^2$
 $A_c = \pi * (d - 2 * t)^2 / 400 = 532,56 \text{ cm}^2$
 $I_c = \pi / 640000 * (d - 2 * t)^4 = 22570,10 \text{ cm}^4$

Local buckling of parts of the steel section:

$$\varepsilon = \sqrt{(235 / f_{yk})} = 0,81$$

$$(d/t) / (90 * \varepsilon^2) = \underline{0,73 < 1}$$

Structural verification of the column's load-bearing capacity:

$$N_{pl,Rd} = (A_a * f_{yd} + A_c * f_{cd}) / 10 = 2768,81 \text{ kN}$$

Structural verification procedure applicable?

$$\delta = A_a * f_{yk} / \gamma_a / (N_{pl,Rd} * 10) = 0,62$$

$$0,2 / \delta = \underline{0,32 < 1}$$

$$\delta / 0,9 = \underline{0,69 < 1}$$

⇒ Structural verification procedure applicable

$$N_{pl,Rk} = (A_a * f_{yk} + A_c * \alpha * f_{ck}) / 10 = 3471,72 \text{ kN}$$

Effective elastic bending rigidity:

$$EI_e = (E_a * I_a + 0,8 * E_{cd} * I_c) / 10^5 = 14141,18 \text{ kNm}^2$$

Reference degree of slenderness:

$$N_{cr} = \pi^2 * EI_e / h^2 = 2848,32 \text{ kN}$$

$$\lambda' = \sqrt{(N_{pl,Rk} / N_{cr})} = 1,10$$

$$\lambda'_g = 0,8 / (1 - \delta) = 2,11$$

$$\lambda' / \lambda'_g = 0,52 < 1$$

⇒ Long-term behaviour (creep and shrinkage) may be disregarded.

Concrete-filled hollow section, therefore strut curve a

$$\text{Type} = \text{SEL}(\text{"comp/buck"}; \text{Desc};) = \text{concrete-filled hollow section}$$

$$\text{line} = \text{TAB}(\text{"comp/buck"}; \text{line}; \text{Desc}=\text{Type}) = \text{a}$$

$$\kappa = \text{TAB}(\text{"comp/buck"}; \kappa; \text{line}=\text{line}; \lambda_{lat}=\lambda') = 0,596$$

$$N_{Rd} = \kappa * N_{pl,Rd} = 1650,21 \text{ kN}$$

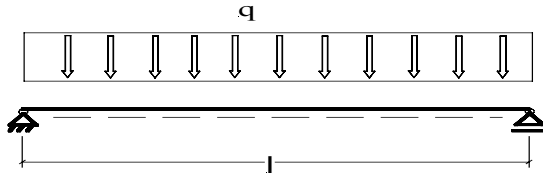
Structural verification:

$$N_{sd} / N_{Rd} = \underline{0,95 < 1}$$

The ultimate load bearing capacity is almost completely exploited.

Floors

Composite beams - Ultimate load bearing capacity:



System:

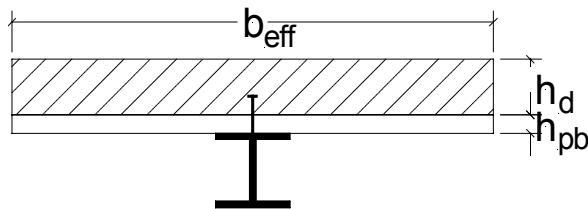
Beam length $L = 12,00$ m
 Beam distance $s = 3,60$ m

Materials:

Beam type = SEL("steel/profils"; Name;) = IPE
 Nominal height $NH =$ SEL("steel/"type; NH;) = 450
 $A =$ TAB("steel/"type; A; NH=NH) = 98,80 cm²
 $h =$ TAB("steel/"type; h; NH=NH)/10 = 45,00 mm
 $I_y =$ TAB("steel/"type; I_y ; NH=NH) = 33740,00 cm⁴
 Shear connectors $\varnothing 22$
 $h_b = 100,00$ mm
 $d = 22,00$ mm
 Distance of shear connectors $e_L = 15,00$ cm
 Concrete = SEL("concrete/EC"; Name;) = C25/30
 Steel = SEL("steel/EC"; NameEN;) = S355

Load:

$g_{1k} = 15,28$ kN/m
 $g_{2k} = 7,74$ kN/m (after removal of temporary props)
 $q_k = 18,00$ kN/m



Floor thickness $h_d = 16,00$ cm
 Thickness of profile steel sheeting $h_{pb} = 5,10$ cm

Material properties:

$E_{cm} =$ TAB("concrete/EC"; E_{cm} ; Name=Concrete) = 30500,00 N/mm²
 $E_a =$ TAB("steel/EC"; E; NameEN=Steel) = 210000,00 N/mm²

Cross section load bearing capacity at full shear connection:

$b_{eff} = 2 * L / 8 = 3,00$ m
 $b_{eff} = \text{MIN}(b_{eff}; s) = \underline{\underline{3,00}} \text{ m}$

Working state analysis of composite beams:

Form factors:

Short term load at point of time $t=0$:

$$n_0 = E_a / E_{cm} = 6,89$$

Effective component thickness:

$$\text{Beam width in air } u = 360,00 \text{ cm}$$

$$h_o = 2 * h_d * u / u = 32,00 \text{ cm}$$

Final creep value $t_0 = 14$ days

$$\text{according to EC2 Appendix1 } \phi = 2,70$$

$$\text{constant continuous load } \psi_{A,B} = 1,10$$

$$\text{Shrinkage } \psi_{A,S} = 0,55$$

$$\text{Creep under continuous load } n\phi = n_0 * (1 + \psi_{A,B} * \phi) = 27,35$$

$$\text{Shrinkage } n_S = n_0 * (1 + \psi_{A,S} * \phi) = 17,12$$

$$A_c = 100 * b_{eff} * (h_d - h_{pb}) = 3270,00 \text{ cm}^2$$

$$I_c = 100 * b_{eff} * (h_d - h_{pb})^3 / 12 = 32375,72 \text{ cm}^2\text{m}^2$$

$$z'_{s0} = (A * (h/2 + h_d) + A_c / n_0 * (h_d - h_{pb}) / 2) / (A + A_c / n_0) = 11,14 \text{ cm}$$

$$(I_y + I_c / n_0) / 10000 + (A * (h/2 + h_d)^2 + A_c / n_0 * ((h_d - h_{pb}) / 2)^2) / 10000 = 19,90 \text{ cm}^2\text{m}^2$$

$$((z'_{s0} / 100)^2 * (A + A_c / n_0) * (-1)) = -7,12 \text{ cm}^2\text{m}^2$$

$$I_{i0} = \underline{\underline{12,78 \text{ cm}^2\text{m}^2}}$$

$$z'_{s\phi} = (A * (h/2 + h_d) + A_c / n_{\phi} * (h_d - h_{pb}) / 2) / (A + A_c / n_{\phi}) = 20,40 \text{ cm}$$

$$(I_y + I_c / n_{\phi}) / 10000 + (A * (h/2 + h_d)^2 + A_c / n_{\phi} * ((h_d - h_{pb}) / 2)^2) / 10000 = 18,49 \text{ cm}^2\text{m}^2$$

$$((z'_{s\phi} / 100)^2 * (A + A_c / n_{\phi}) * (-1)) = -9,09 \text{ cm}^2\text{m}^2$$

$$I_{i\phi} = \underline{\underline{9,40 \text{ cm}^2\text{m}^2}}$$

$$z'_{sS} = (A * (h/2 + h_d) + A_c / n_S * (h_d - h_{pb}) / 2) / (A + A_c / n_S) = 16,72 \text{ cm}$$

$$(I_y + I_c / n_S) / 10000 + (A * (h/2 + h_d)^2 + A_c / n_S * ((h_d - h_{pb}) / 2)^2) / 10000 = 18,78 \text{ cm}^2\text{m}^2$$

$$((z'_{sS} / 100)^2 * (A + A_c / n_S) * (-1)) = -8,10 \text{ cm}^2\text{m}^2$$

$$I_{iS} = \underline{\underline{10,68 \text{ cm}^2\text{m}^2}}$$

Calculation of midspan deflection:Release of temporary prop at point of time $t=0$

$$B = g_{1k} * 1,25 * L / 2 = 114,60 \text{ kN}$$

$$f_{B0} = 100 * B * L^3 / (48 * E_a / 10 * I_{i0}) = 1,54 \text{ cm}$$

Assumption: 40% of the live load as a permanent load at point of time $t=0$:

$$\text{Ratio} = 0,40$$

$$f_{g2,0} = 100 * 5/384 * (g_{2k} + \text{Ratio} * q_k) * L^4 / (E_a / 10 * I_{i0}) = 1,50 \text{ cm}$$

from imposed load (short term ratio):

$$\text{Ratio2} = 1 - \text{Ratio} = 0,60$$

$$f_q = 100 * 5/384 * \text{Ratio2} * q_k * L^4 / (E_a / 10 * I_{i0}) = 1,09 \text{ cm}$$

From removal of temporary prop at point of time $t = \infty$:

$$f_B = f_{B0} * I_{i0} / I_{i\Phi} = 2,09 \text{ cm}$$

From $g_{k2} + q_{\text{permanent}}$ at point of time $t = \infty$:

$$f_{g2} = f_{g2,0} * I_{i0} / I_{i\Phi} = 2,04 \text{ cm}$$

From shrinkage at point of time $t = \infty$:

$$\text{Final shrinkage value } \epsilon_{cS} = 325 * 10^{-6}$$

$$N_{\text{sch}} = (A_c * E_a / n_s * \epsilon_{cS}) / 10 = 1303,61 \text{ kN}$$

$$Z_{\text{sch}} = z'_{sS} - (h_d - h_{pb}) / 2 = 11,27 \text{ cm}$$

$$M_{\text{sch}} = N_{\text{sch}} * Z_{\text{sch}} / 100 = 146,92 \text{ kN}$$

$$f_{\text{sch}} = 100 * 1/8 * M_{\text{sch}} * L^2 / (E_a / 10 * I_{iS}) = 1,18 \text{ cm}$$

Maximum deflection:

$$\text{max}_f = f_B + f_{g2} + f_{\text{sch}} + f_q = 6,40 \text{ cm}$$

Recommended camber:

$$f_0 = f_{B0} + f_{g2,0} = 3,04 \text{ cm}$$

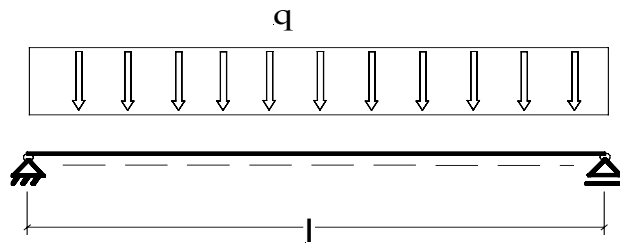
Maximum deflection in final state:

$$f = \text{max}_f - f_0 = 3,36 \text{ cm}$$

Structural verification:

$$f / 100 / (L / 300) = \underline{\underline{0,84 < 1}}$$

The deflection in the working state meets the required standard!

Composite floor in the final state:**System:**

Steel profile with lugs; values according to manufacturer's specifications (approval)

$m =$	166,00
$k =$	0,15
Sheet thickness $t =$	0,86 mm
$e =$	17,00 mm
Beam length $L =$	4,80 m
Stability $f_{yp} =$	350,00 N/mm ²
Cross section area $A_p =$	1562,00 N/mm ²
$\tau_{u,Rd} =$	280,00 kN/m
minimum width of concrete ribs $b_0 =$	750,00 mm/m

Concrete:

Floor thickness $h_t =$	14,00 cm
Concrete =	SEL("concrete/EC"; Name;) = C25/30
$f_{ck} =$	TAB("concrete/EC"; fck; Name=Concrete) = 25,00 N/mm ²
$f_{ctk005} =$	TAB("concrete/EC"; fctk05; Name=Concrete) = 1,80 N/mm ²

Influences:

Final state:

Permanent load of composite floor $G_1 =$	3,30 kN/m
Permanent load of design loads $G_2 =$	1,20 kN/m
Imposed load $Q =$	5,00 kN/m

Partial safety factors:

Dead load $\gamma_G =$	1,35
Imposed Load $\gamma_Q =$	1,50
Concrete $\gamma_c =$	1,50
Profile steel sheeting $\gamma_{ap} =$	1,10
Longitudinal shear $\gamma_{vs} =$	1,25

Structural verification of the composite floor in the final state:

$$b = 1000,00 \text{ mm}$$

Deflection:

$$M_{Sd} = (\gamma_G * (G_1 + G_2) + \gamma_Q * Q) * L^2 / 8 = 39,10 \text{ kNm/m}$$

Design resistance:

$$N_{cf} = (A_p * f_{yp} / \gamma_{ap}) / 1000 = 497,00 \text{ kN}$$

$$\alpha = 0,85$$

$$\chi = 1000 * N_{cf} / (b * (\alpha * f_{ck} / \gamma_c)) = 35,08$$

$$d_p = 10 * h_t - e = 123,00 \text{ mm}$$

$$M_{p,Rd} = N_{cf} * (d_p - 0,5 * \chi) / 1000 = 52,41 \text{ kNm/m}$$

Structural verification:

$$M_{Sd} / M_{p,Rd} = \underline{\underline{0,75 < 1}}$$

Longitudinal shear, m + k - method:

Design shear force at the footing point:

$$V_{Sd} = (\gamma_G * (G_1 + G_2) + \gamma_Q * Q) * L / 2 = 32,58 \text{ kN/m}$$

Longitudinal shear resistance:

$$\text{Shear length } L_s = 1000 * L / 4 = 1200,00 \text{ mm}$$

$$V_{I,Rd} = (b * d_p * (m * A_p / (b * L_s) + k) / \gamma_{Vs}) / 1000 = 36,02 \text{ kN/m}$$

Structural verification:

$$V_{Sd} / V_{I,Rd} = \underline{0,90 < 1}$$

Longitudinal shear, partial shear connection (EC4, Appendix E):

Design value of the bond strength according to manufacturer's information:

$$\text{Shear length at full shear connection } \eta = 1,00$$

$$L_{sf} = N_{cf} / (b / 1000 * \tau_{u,Rd}) = 1,77 \text{ m}$$

Design resistance:

$$\tau_{Rd} = 0,09 * (f_{ck})^{(1/3)} = 0,26 \text{ N/mm}^2$$

$$k_v = 1,6 - d_p / 1000 = 1,48 > 1$$

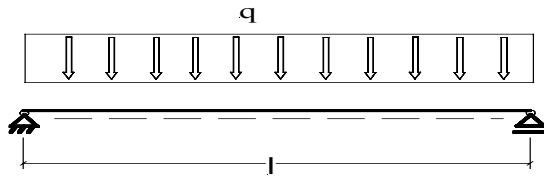
$$A_p = b_0 * t = 645,00 \text{ mm}^2$$

$$\rho = A_p / (b_0 * d_p) = 0,007$$

$$V_{v,Rd} = (b_0 * d_p * \tau_{Rd} * k_v * (1,2 + 40 * \rho)) / 1000 = 52,54 \text{ kN/m}$$

Structural verification:

$$V_{Sd} / V_{v,Rd} = \underline{0,62 < 1}$$

Composite beams - Ultimate load bearing capacity:**System:**

Beam length $L =$	12,00 m
Number of temporary props $a =$	1,00
Beam distance $s =$	3,60 m
Flange width $b_0 =$	12,60 cm

Materials:

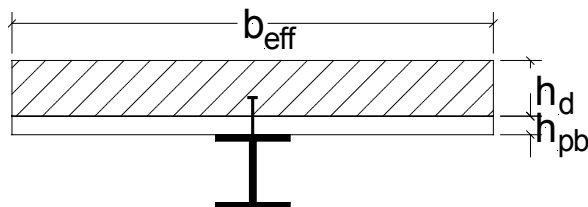
Concrete =	SEL("concrete/EC"; Name;)	=	C25/30
Steel =	SEL("steel/EC"; NameEN;)	=	S355
Beam type =	SEL("steel/profils"; Name;)	=	IPE
Nominal height $NH =$	SEL("steel/type; NH;)	=	400
$A =$	TAB("steel/type; A; NH=NH)	=	84,50 cm ²
$h =$	TAB("steel/type; h; NH=NH)/10	=	40,00 cm
$t =$	TAB("steel/type; s; NH=NH)/10	=	0,86 cm
$I_y =$	TAB("steel/type; I_y ; NH=NH)	=	23130,00 cm ⁴
$M_{pl} =$	TAB("steel/type; M_{plyd} ; NH=NH)	=	289,00 kNm
$M_{pl} =$	IF(Steel="S355";1,5;1)* M_{pl}	=	433,50 kNm
Shear connectors $\varnothing 22$			
$h_b =$		=	100,00 mm
$d =$		=	19,00 mm
Distance of shear connectors $e_L =$		=	15,00 cm
Ultimate strength $f_u =$		=	450,00 N/mm ²

Load:

$g_k =$	14,50 kN/m
$q_k =$	18,00 kN/m

Partial safety factors:

Dead load $\gamma_G =$	1,35
Imposed Load $\gamma_Q =$	1,50
Concrete $\gamma_c =$	1,50
Construction steel $\gamma_a =$	1,10
Profile steel sheeting $\gamma_{ap} =$	1,10
Longitudinal shear $\gamma_{vs} =$	1,25



Floor thickness $h_d =$	16,00 cm
Thickness of profile steel sheeting $h_{pb} =$	5,10 cm
Cross-sectional area of plate $A_p =$	15,62 kN/m

Material properties:

$$\begin{aligned}
 E_{cm} &= \text{TAB}(\text{"concrete/EC"}; E_{cm}; \text{Name=Concrete}) &= & 30500,00 \text{ N/mm}^2 \\
 f_{ck} &= \text{TAB}(\text{"concrete/EC"}; f_{ck}; \text{Name=Concrete}) &= & 25,00 \text{ N/mm}^2 \\
 f_{ctk005} &= \text{TAB}(\text{"concrete/EC"}; f_{ctk05}; \text{Name=Concrete}) &= & 1,80 \text{ N/mm}^2 \\
 f_{cd} &= f_{ck} / \gamma_c &= & 16,67 \text{ N/mm}^2 \\
 \\
 E_a &= \text{TAB}(\text{"steel/EC"}; E; \text{NameEN=Steel}) &= & 210000,00 \text{ N/mm}^2 \\
 f_{yk} &= \text{TAB}(\text{"steel/EC"}; f_y; \text{NameEN=Steel}) &= & 355,00 \text{ N/mm}^2 \\
 f_{yd} &= f_{yk} / \gamma_a &= & 322,73 \text{ N/mm}^2
 \end{aligned}$$

Holorib sheeting in accordance with building regulations approval:

$$f_{yp} = 280,00 \text{ N/mm}^2$$

Conformation of fitness for purpose:

Stress resultants in ultimate state analysis of fitness for purpose:

$$\begin{aligned}
 r_d &= \gamma_G * g_k + \gamma_Q * q_k &= & 46,58 \text{ kN/m} \\
 M_{Sd} &= r_d * L^2 / 8 &= & 838,44 \text{ kNm} \\
 V_{Sd} &= r_d * L / 2 &= & 279,48 \text{ kN}
 \end{aligned}$$

Cross section load bearing capacity with full shear connection:

$$\begin{aligned}
 b_{eff} &= 2 * L / 8 &= & 3,00 \text{ m} \\
 b_{eff} &= \text{MIN}(b_{eff}; s) &= & \underline{3,00 \text{ m}} \\
 A_v &= 1.04 * h * t &= & 35,78 \text{ cm}^2 \\
 N_{pl,a,Rd} &= A * f_{yd} / 10 &= & 2727,07 \text{ kN} \\
 N_{cd} &= 0.85 * f_{cd} * 100 * b_{eff} * (h_d - h_{pb}) &= & 46334,26 \text{ kN} \\
 \alpha_c &= 0,85 \\
 z_{pl} &= N_{pl,a,Rd} / (10 * \alpha_c * f_{cd} * b_{eff}) &= & 6,42 \text{ cm}
 \end{aligned}$$

⇒ The plastic neutral axis is in the concrete chord above the profile steel sheeting.

Distance to centre of gravity:

$$\begin{aligned}
 a &= h / 2 + h_d - z_{pl} / 2 &= & 32,79 \text{ cm} \\
 \Rightarrow M_{pl,Rd} &= N_{pl,a,Rd} * a / 100 &= & 894,21 \text{ kNm} \\
 V_{pl,Rd} &= A_v * f_{yk} / (10 * \sqrt{3} * \gamma_a) &= & 666,68 \text{ kN}
 \end{aligned}$$

Structural verifications:

$$\begin{aligned}
 M_{Sd} / M_{pl,Rd} &= &= & \underline{0,94 < 1} \\
 V_{Sd} / V_{pl,Rd} &= &= & \underline{0,42 < 1}
 \end{aligned}$$

Longitudinal shear load capacity and shear connection:

$$\begin{aligned}
 h_b / d &= &= & 5,26 > 4 \rightarrow \text{shear connectors are ductile} \\
 \alpha &= \text{IF}(h_b / d > 4; 1; 0.2 * ((h_b / d) + 1)) &= & 1,00
 \end{aligned}$$

Marginal force of a shear connector:

$$\begin{aligned}
 P_{Rd1} &= (0.8 * f_u * \pi * d^2 / (4 * \gamma_{vs})) / 1000 &= & 81,66 \text{ kN} \\
 P_{Rd2} &= (0,29 * \alpha * d^2 * \sqrt{(f_{ck} * E_{cm}) / \gamma_{vs}}) / 1000 &= & 73,13 \text{ kN} \\
 P_{Rd} &= \text{MIN}(P_{Rd1}; P_{Rd2}) &= & 73,13 \text{ kN}
 \end{aligned}$$

single-row:

$$N_r = 1,00 \text{ (one row of shear connectors)}$$

$$k_{t1} = (0.7 / \sqrt{N_r}) * (b_0 / h_{pb}) * ((h_b / 10) / h_{pb} - 1) = 1,66$$

$$k_{t2} = 0,85$$

$$k_t = \text{MIN} (k_{t1}; k_{t2}) = 0,85$$

$$P_{Rd1} = k_t * P_{Rd} = 62,16 \text{ kN}$$

double-row:

$$N_r = 2,00 \text{ (two rows of shear connectors)}$$

$$k_{t1} = (0.7 / \sqrt{N_r}) * (b_0 / h_{pb}) * ((h_b / 10) / h_{pb} - 1) = 1,17$$

$$k_{t2} = 0,70$$

$$k_t = \text{MIN} (k_{t1}; k_{t2}) = 0,70$$

$$P_{Rd2} = k_t * P_{Rd} = 51,19 \text{ kN}$$

Possible shear connector configurations:

$$7 \text{ pairs of shear connectors at the end of the beam Anzp} = 7,00$$

$$33 \text{ individual shear connectors in the span Anze} = 33,00$$

$$\text{aufn}_V = \text{Anze} * P_{Rd1} + 2 * \text{Anzp} * P_{Rd2} = 2767,94 \text{ kN}$$

Structural verification:

$$N_{pl,a,Rd} / \text{aufn}_V = \underline{\underline{0,99 < 1}}$$

Structural verification against diagonal strut failure according to NAD for Germany:

$$A_{cv} = 100 * (h_d - h_{pb}) = 1090,00 \text{ N/m}$$

$$\text{Normal concrete } \eta = 1,00$$

$$v_{Rd,2} = 0.2 * A_{cv} * \eta * f_{ck} / (\gamma_c * 10) = 363,33 \text{ kN/m}$$

$$v_{Sd} = P_{Rd1} / (e_L / 100) = 414,40 \text{ kN/m}$$

$$v_{Sd,li} = v_{Sd} * 1/2 = 207,20 \text{ kN/m}$$

$$v_{Sd,li} / v_{Rd,2} = \underline{\underline{0,57 < 1}}$$

This calculation disregards the input from profile steel sheeting!

Cross reinforcement necessary:

$$\text{NAD } \tau_{Rd} = 0.09 * f_{ck}^{(1/3)} = 0,26 \text{ N/mm}^2$$

$$v_{pd} = A_p * f_{yp} / (\gamma_{ap} * 10) = 397,60 \text{ kN/m}$$

$$v_{Rd,3} = 2.5 * A_{cv} * \eta * (\tau_{Rd} / 10) + v_{pd} = 468,45 \text{ kN/m}$$

Structural verification of diagonal strut:

$$v_{Sd,li} / v_{Rd,3} = \underline{\underline{0,44 < 1}}$$

Working state analysis of composite beams:

Form factors:

Short term load at point of time t=0:

$$n_0 = E_a / E_{cm} = 6,89$$

Effective component thickness:

$$\text{Beam width in air } u = 360,00 \text{ cm}$$

$$h_o = 2 * h_d * u / u = 32,00 \text{ cm}$$

Final creep value $t_0 = 14$ days

according to EC2 Appendix1 $\phi =$

$$\begin{aligned} \text{Creep under continuous load } n_{\phi} &= n_0 * 3 &= 20,67 \\ \text{Shrinkage and creep } n_S &= n_0 * 2 &= 13,78 \\ A_c &= 100 * b_{\text{eff}} * (h_d - h_{\text{pb}}) &= 3270,00 \text{ cm}^2 \\ I_c &= 100 * b_{\text{eff}} * (h_d - h_{\text{pb}})^3 / 12 &= 32375,72 \text{ cm}^2\text{m}^2 \end{aligned}$$

$$z'_{s0} = (A * (h/2 + h_d) + A_c / n_0 * (h_d - h_{\text{pb}}) / 2) / (A + A_c / n_0) = 10,07 \text{ cm}$$

$$\begin{aligned} (I_y + I_c / n_0) / 10000 + (A * (h/2 + h_d)^2 + A_c / n_0 * ((h_d - h_{\text{pb}}) / 2)^2) / 10000 &= 15,14 \text{ cm}^2\text{m}^2 \\ ((z'_{s0} / 100)^2 * (A + A_c / n_0) * (-1)) &= -5,67 \text{ cm}^2\text{m}^2 \end{aligned}$$

$$I_{i0} = \underline{\underline{9,47 \text{ cm}^2\text{m}^2}}$$

$$z'_{s\phi} = (A * (h/2 + h_d) + A_c / n_{\phi} * (h_d - h_{\text{pb}}) / 2) / (A + A_c / n_{\phi}) = 16,09 \text{ cm}$$

$$\begin{aligned} (I_y + I_c / n_{\phi}) / 10000 + (A * (h/2 + h_d)^2 + A_c / n_{\phi} * ((h_d - h_{\text{pb}}) / 2)^2) / 10000 &= 13,89 \text{ cm}^2\text{m}^2 \\ ((z'_{s\phi} / 100)^2 * (A + A_c / n_{\phi}) * (-1)) &= -6,28 \text{ cm}^2\text{m}^2 \end{aligned}$$

$$I_{i\phi} = \underline{\underline{7,61 \text{ cm}^2\text{m}^2}}$$

$$z'_{sS} = (A * (h/2 + h_d) + A_c / n_S * (h_d - h_{\text{pb}}) / 2) / (A + A_c / n_S) = 13,47 \text{ cm}$$

$$\begin{aligned} (I_y + I_c / n_S) / 10000 + (A * (h/2 + h_d)^2 + A_c / n_S * ((h_d - h_{\text{pb}}) / 2)^2) / 10000 &= 14,20 \text{ cm}^2\text{m}^2 \\ ((z'_{sS} / 100)^2 * (A + A_c / n_S) * (-1)) &= -5,84 \text{ cm}^2\text{m}^2 \end{aligned}$$

$$I_{iS} = \underline{\underline{8,36 \text{ cm}^2\text{m}^2}}$$

Calculation of midspan deflection:

Release of temporary prop at point of time $t = 0$

$$\begin{aligned} B &= g_k * 1,25 * L / 2 &= 108,75 \text{ kN} \\ f_{B0} &= 100 * B * L^3 / (48 * E_a / 10 * I_{i0}) &= 1,97 \text{ cm} \end{aligned}$$

Assumption: 40% of the live load as a permanent load at point of time $t = 0$:

$$\begin{aligned} \text{Ratio} &= 0,40 \\ f_{g2,0} &= 100 * 5/384 * \text{Ratio} * q_k * L^4 / (E_a / 10 * I_{i0}) &= 0,98 \text{ cm} \end{aligned}$$

$$f_0 = f_{B0} + f_{g2,0} = 2,95 \text{ cm}$$

from imposed load (short term ratio):

$$\begin{aligned} \text{Ratio2} &= 1 - \text{Ratio} &= 0,60 \\ f_q &= 100 * 5/384 * \text{Ratio2} * q_k * L^4 / (E_a / 10 * I_{i0}) &= 1,47 \text{ cm} \end{aligned}$$

From removal of temporary prop at point of time $t = \infty$:

$$f_B = f_{B0} * I_{i0} / I_{i\phi} = 2,45 \text{ cm}$$

From $g_{k2} + q_{\text{permanent}}$ at point of time $t = \infty$:

$$f_{g2} = f_{g2,0} * I_{i0} / I_i \Phi = 1,22 \text{ cm}$$

$$f_d = f_B + f_{g2} = 3,67 \text{ cm}$$

Creep under continuous load only:

$$f_k = f_d - f_0 = 0,72 \text{ cm}$$

From shrinkage at point of time $t = \infty$:

$$\text{Final shrinkage value } \epsilon_{cS} = 325 * 10^{-6}$$

$$N_{sch} = (A_c * E_a / n_S * \epsilon_{cS}) / 10 = 1619,58 \text{ kN}$$

$$z_{sch} = z'_{sS} - (h_d - h_{pb}) / 2 = 8,02 \text{ cm}$$

$$M_{sch} = N_{sch} * z_{sch} / 100 = 129,89 \text{ kN}$$

$$f_{sch} = 100 * 1/8 * M_{sch} * L^2 / (E_a / 10 * I_{iS}) = 1,33 \text{ cm}$$

Maximum deflection:

$$\max_f = f_B + f_{g2} + f_{sch} + f_q = 6,47 \text{ cm}$$

Recommended camber:

$$f_0 = f_{B0} + f_{g2,0} = 2,95 \text{ cm}$$

Maximum deflection in final state:

$$f = \max_f - f_0 = 3,52 \text{ cm}$$

Structural verification:

$$f / 100 / (L / 250) = \underline{0,73 < 1}$$

The deflection in the working state meets the required standard!

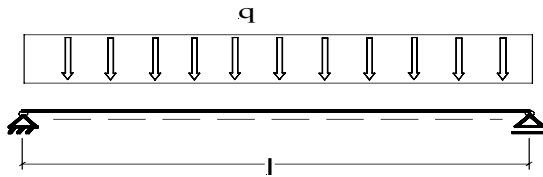
Vibration behaviour:

$$\omega = 10 * \pi^2 / L^2 * \sqrt{(E_a * I_{i0} / (g_k * 98.1))} = 25,63 \text{ (1/s)}$$

$$f = \omega / (2 * \pi) = 4,08 \text{ Hz}$$

The natural frequency should be no less than 3 Hz.

For sports or dance halls, it should be no less than 5 Hz.

POS.: Composite beams - Ultimate load bearing capacity:**System:**

Beam length $L =$	12,00 m
Number of temporary props $a =$	1,00
Beam distance $s =$	3,60 m
Flange width $b_0 =$	12,60 cm

Materials:

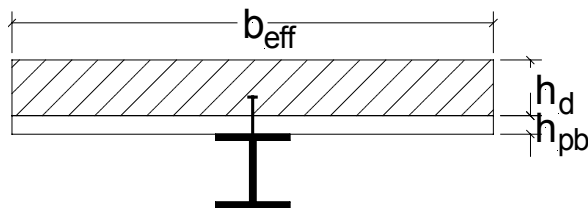
Concrete =	SEL("concrete/EC"; Name;)	=	C25/30
Steel =	SEL("steel/EC"; NameEN;)	=	S355
Beam type =	SEL("steel/profils"; Name;)	=	IPE
Nominal height $NH =$	SEL("steel/type; NH;)	=	450
$A =$	TAB("steel/type; A; NH=NH)	=	98,80 cm ²
$h =$	TAB("steel/type; h; NH=NH)/10	=	45,00 cm
$t =$	TAB("steel/type; s; NH=NH)/10	=	0,94 cm
$I_y =$	TAB("steel/type; I_y ; NH=NH)	=	33740,00 cm ⁴
$M_{pl} =$	TAB("steel/type; M_{plyd} ; NH=NH)	=	373,00 kNm
$M_{pl} =$	IF(Steel="S355";1,5;1)* M_{pl}	=	559,50 kNm
Shear connectors $\varnothing 22$			
$h_b =$	100,00 mm		
$d =$	22,00 mm		
Distance of shear connectors $e_L =$	15,00 cm		
Ultimate strength $f_u =$	450,00 N/mm ²		

Load:

$g_{k1} =$	15,28 kN/m
$g_{k2} =$	7,74 kN/m (after removal of temporary props)
$q_k =$	18,00 kN/m

Partial safety factors:

Dead load $\gamma_G =$	1,35
Imposed Load $\gamma_Q =$	1,50
Concrete $\gamma_c =$	1,50
Construction steel $\gamma_a =$	1,10
Profile steel sheeting $\gamma_{ap} =$	1,10
Longitudinal shear $\gamma_{vs} =$	1,25



Floor thickness $h_d =$	16,00 cm
Thickness of profile steel sheeting $h_{pb} =$	5,10 cm
Cross-sectional area of plate $A_p =$	15,62 kN/m

Material properties:

$$\begin{aligned}
 E_{cm} &= \text{TAB}(\text{"concrete/EC"}; E_{cm}; \text{Name=Concrete}) &= & 30500,00 \text{ N/mm}^2 \\
 f_{ck} &= \text{TAB}(\text{"concrete/EC"}; f_{ck}; \text{Name=Concrete}) &= & 25,00 \text{ N/mm}^2 \\
 f_{ctk005} &= \text{TAB}(\text{"concrete/EC"}; f_{ctk05}; \text{Name=Concrete}) &= & 1,80 \text{ N/mm}^2 \\
 f_{cd} &= f_{ck} / \gamma_c &= & 16,67 \text{ N/mm}^2
 \end{aligned}$$

$$\begin{aligned}
 E_a &= \text{TAB}(\text{"steel/EC"}; E; \text{NameEN=Steel}) &= & 210000,00 \text{ N/mm}^2 \\
 f_{yk} &= \text{TAB}(\text{"steel/EC"}; f_y; \text{NameEN=Steel}) &= & 355,00 \text{ N/mm}^2 \\
 f_{yd} &= f_{yk} / \gamma_a &= & 322,73 \text{ N/mm}^2
 \end{aligned}$$

Holorib sheeting in accordance with building regulations approval:

$$f_{yp} = 280,00 \text{ N/mm}^2$$

Conformation of fitness for purpose:

Stress resultants in ultimate state analysis of fitness for purpose:

$$\begin{aligned}
 r_d &= \gamma_G * (g_{k1} + g_{k2}) + \gamma_Q * q_k &= & 58,08 \text{ kN/m} \\
 M_{Sd} &= r_d * L^2 / 8 &= & 1045,44 \text{ kNm} \\
 V_{Sd} &= r_d * L / 2 &= & 348,48 \text{ kN}
 \end{aligned}$$

Cross section load bearing capacity with full shear connection:

$$\begin{aligned}
 b_{eff} &= 2 * L / 8 &= & 3,00 \text{ m} \\
 b_{eff} &= \text{MIN}(b_{eff}; s) &= & \underline{3,00 \text{ m}}
 \end{aligned}$$

$$A_v = 1.04 * h * t = 43,99 \text{ cm}^2$$

$$N_{pl,a,Rd} = A * f_{yd} / 10 = 3188,57 \text{ kN}$$

$$N_{cd} = 0.85 * f_{cd} * 10 * b_{eff} * (h_d - h_{pb}) = 4633,43 \text{ kN}$$

$$\alpha_c = 0,85$$

$$z_{pl} = N_{pl,a,Rd} / (10 * \alpha_c * f_{cd} * b_{eff}) = 7,50 \text{ cm}$$

⇒ The plastic neutral axis is in the concrete chord above the profile steel sheeting.

Distance to centre of gravity:

$$a = h / 2 + h_d - z_{pl} / 2 = 34,75 \text{ cm}$$

$$\Rightarrow M_{pl,Rd} = N_{pl,a,Rd} * a / 100 = 1108,03 \text{ kNm}$$

$$V_{pl,Rd} = A_v * f_{yk} / (10 * \sqrt{3} * \gamma_a) = 819,65 \text{ kN}$$

Structural verifications:

$$M_{Sd} / M_{pl,Rd} = \underline{0,94 < 1}$$

$$V_{Sd} / V_{pl,Rd} = \underline{0,43 < 1}$$

Longitudinal shear capacity and shear connection:

$$h_b / d = 4,55 > 4 \rightarrow \text{shear connectors are ductile}$$

$$\alpha = \text{IF}(h_b / d > 4; 1; 0.2 * ((h_b / d) + 1)) = 1,00$$

Marginal force of a shear connector:

$$P_{Rd1} = (0.8 * f_u * \pi * d^2 / (4 * \gamma_{vs})) / 1000 = 109,48 \text{ kN}$$

$$P_{Rd2} = (0,29 * \alpha * d^2 * \sqrt{(f_{ck} * E_{cm}) / \gamma_{vs}}) / 1000 = 98,05 \text{ kN}$$

$$P_{Rd} = \text{MIN}(P_{Rd1}; P_{Rd2}) = 98,05 \text{ kN}$$

$$\begin{aligned}
 \text{Number of rows of shear connectors } N_r &= 1,00 \\
 k_{t1} &= (0.7 / \sqrt{N_r}) * (b_0 / h_{pb}) * ((h_b / 10) / h_{pb} - 1) = 1,66 \\
 k_{t2} &= 0,75 \\
 k_t &= \text{MIN} (k_{t1} ; k_{t2}) = 0,75 \\
 P_{Rd} &= k_t * P_{Rd} = 73,54 \text{ kN}
 \end{aligned}$$

Required number of shear connectors for full shear connection:

$$\begin{aligned}
 \text{erf}_n &= N_{pl,a,Rd} / P_{Rd} + 0.49 = 44 \text{ Dübel} \\
 \text{vorh}_n &= 100 * L / 2 / e_L + 0.49 = 40 \text{ Dübel} \\
 \text{because: } \text{erf}_n / \text{vorh}_n &= 1,10 \text{ is a partial connection} \\
 \text{The maximum transferrable concrete compression force is:} \\
 N_c &= \text{vorh}_n * P_{Rd} = 2941,60 \text{ kN} \\
 \text{Degree of shear connection } \eta &= N_c / N_{pl,a,Rd} = 0,92 < 1
 \end{aligned}$$

Reduced plastic bending moment, calculated with linear interpolation:

$$M_{Rd} = M_{pl} + \eta * (M_{pl,Rd} - M_{pl}) = 1064,15 \text{ kNm}$$

Structural verification:

$$M_{Sd} / M_{Rd} = \underline{\underline{0,98 < 1}}$$

Note: No reduction in interaction, because maximum moment and maximum lateral shear force do not occur in the same place.

Shear connector configurations permissible?

a) Cross section class 1 or 2 \Rightarrow OK

$$b) \text{erf}_\eta = 0.25 + 0.03 * L = 0,61$$

$$\text{erf}_\eta / \eta = \underline{\underline{0,66 < 1}}$$

c)

$$M_{pl,Rd} / (2.5 * M_{pl}) = \underline{\underline{0,79 < 1}}$$

All conditions a-c have to be met!

Connection of lateral concrete chord:

Lateral shear::

$$v_{Sd} = P_{Rd} / (e_L / 100) = 490,27 \text{ kN/m}$$

Relevant:

$$v_{Sd,li} = v_{Sd} * 1 / 2 = 245,13 \text{ kN/m}$$

$$A_{cv} = (h_d - h_{pb}) * 100 = 1090,00 \text{ cm}^2/\text{m}$$

$$\text{Normal concrete } \eta = 1,00$$

$$v_{Rd,2} = 0.2 * A_{cv} * \eta * f_{ck} / (\gamma_c * 10) = 363,33 \text{ kN/m}$$

Structural verification against diagonal strut failure:

$$v_{Sd,li} / v_{Rd,2} = \underline{\underline{0,67 < 1}}$$

This calculation disregards the input from profile steel sheeting!

$$\text{EC4 } \tau_{Rd1} = 0.25 * f_{ctk005} / \gamma_c = 0,30 \text{ MN/mm}^2$$

$$\text{NAD } \tau_{Rd2} = 0.09 * f_{ck}^{(1/3)} = 0,26 \text{ MN/mm}^2$$

$$\tau_{Rd} = \text{MIN} (\tau_{Rd1} ; \tau_{Rd2}) = 0,26 \text{ MN/mm}^2$$

$$v_{pd} = A_p * f_{yp} / (\gamma_{ap} * 10) = 397,60 \text{ kN/m}$$

$$v_{Rd,3} = 2.5 * A_{cv} * \eta * (\tau_{Rd} / 10) + v_{pd} = 468,45 \text{ kN/m}$$

Structural verification of diagonal strut:

$$v_{Sd,li} / v_{Rd,3} = \underline{0,52 < 1}$$

The cross reinforcement has been ignored in this calculation.

In the construction, provision should be made for 0.2% of the relevant cross section area A_{cv} that may also be taken into account as part of the bending reinforcement to absorb the negative moment (moment at support) of the transversal continuous composite floor.